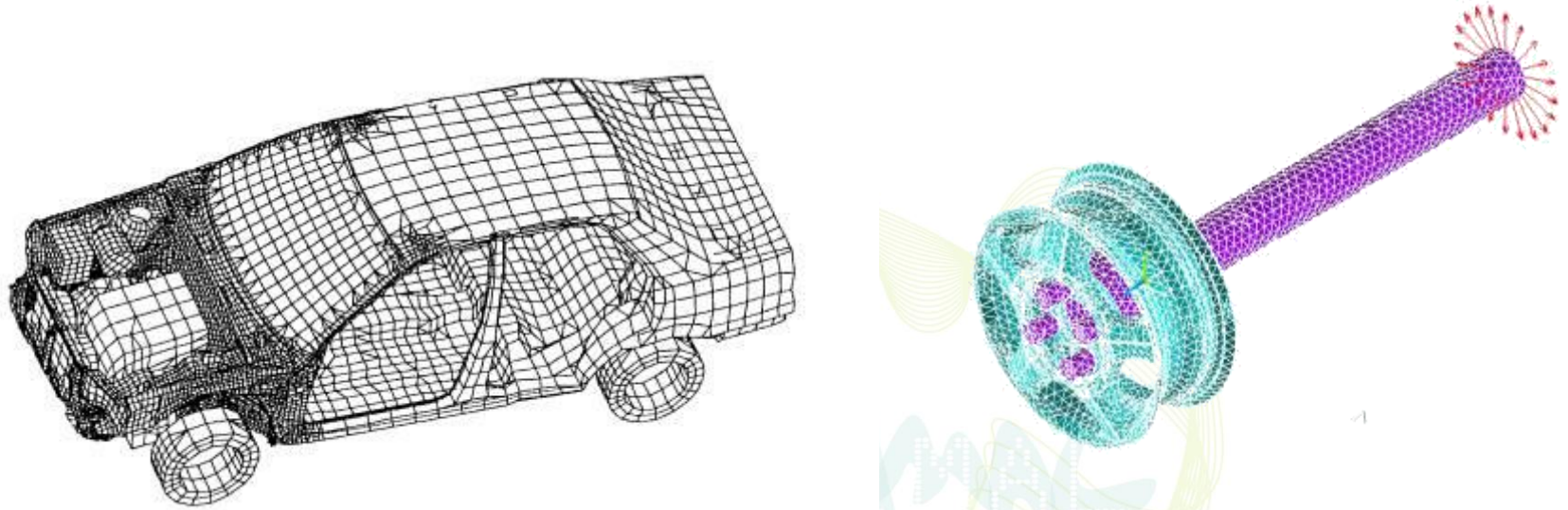


# Chapter 9. Finite Element Analysis

- ✓ In practical mechanical design projects, the fundamental stress analysis equations are often not enough to handle the complicated geometry and boundary conditions.
- ✓ Design and analysis are no longer independent tasks. **Finite Element Analysis** (FEA) has become a fundamental design tool for design engineers. In many commercial software, finite element analysis has been combined with solid modeling to become integrated computer aided design or computer aided engineering software..
- ✓ Finite element analysis is a mathematical method that can be used to solve various problems such as stress analysis, heat conduction, electrical fields, magnetic fields, ideal fluid flow, etc.



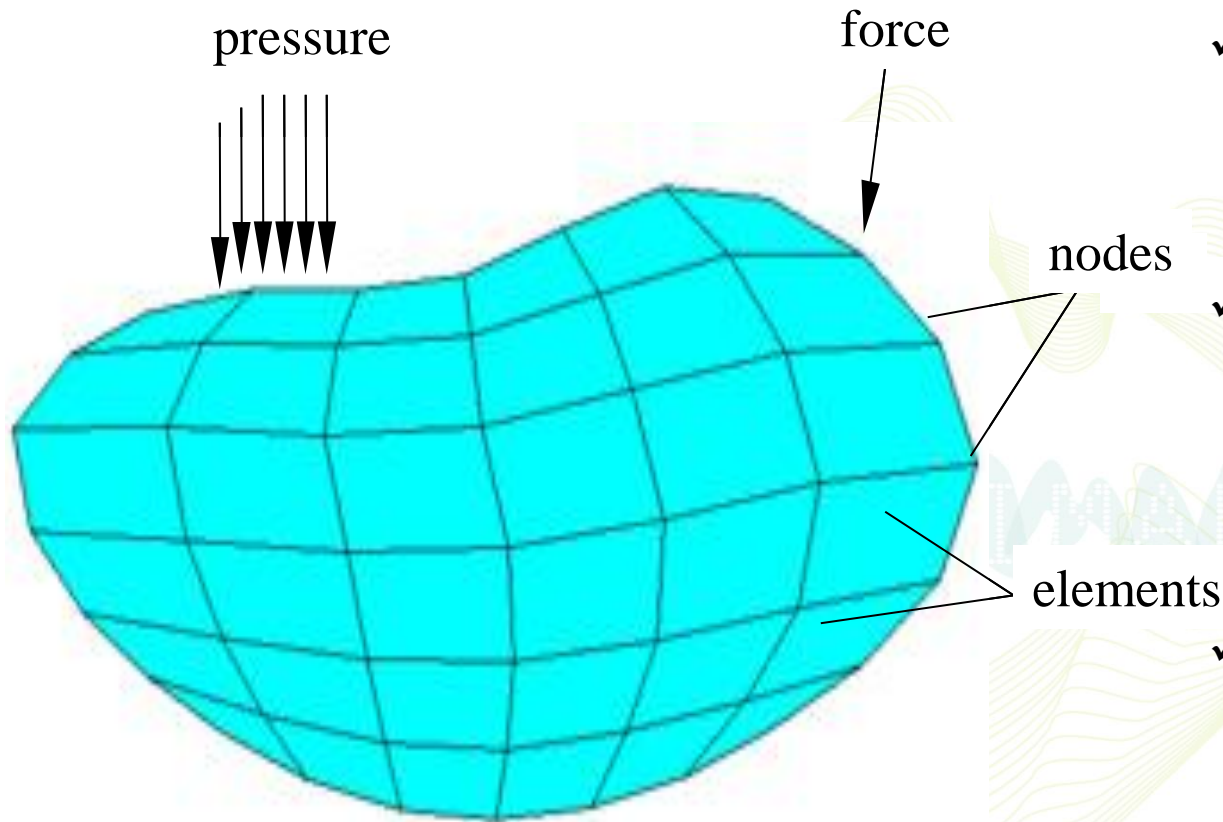
# General Procedure for using an FEA Software



- ✓ In structural analysis, FEA can be used to solve for the displacement and stress for a complicated structure, under a given loading and boundary condition.
- ✓ Pre-processor → Solver → Post-processor
- ✓ Designers can define the **geometry, loading, and boundary conditions** in pre-processor.



# Pre-Processor



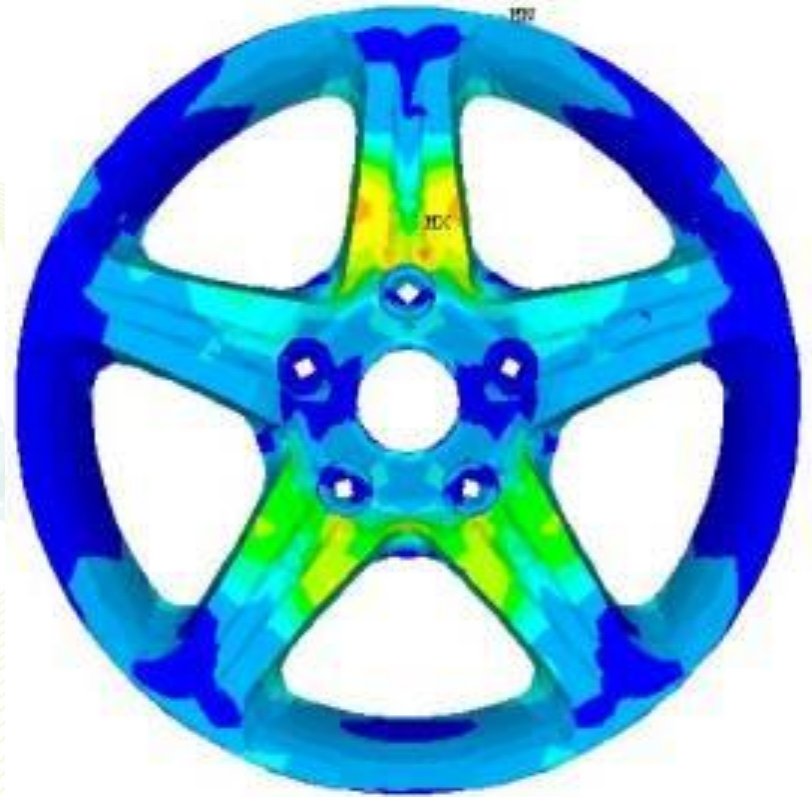
- ✓ In FEA method, an object is divided into a large number of **elements**, which is composed of **nodes**.
- ✓ Designers defines the geometry of the structure by defining the coordinates of nodes and connectivity of elements.
- ✓ Many FEA software provides **automatic mesh generator**.

- ✓ In a finite element model, loading is often forces applied on nodes or pressure applied on the boundary of an element.
- ✓ Boundary conditions are represented by the degrees of freedom of the nodes.

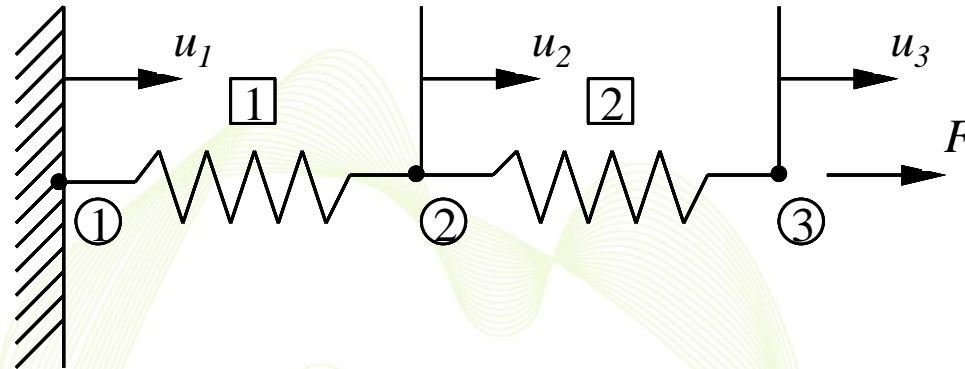


# Solver and Post-Processor

- ✓ In the solver, system equations are constructed according to the nodes, elements, loading, and boundary conditions defined by the designers. The system equations are then solved.
- ✓ The solution data is displayed graphically in the post processor to give the designer a more intuitive feeling.
- ✓ For example, nodal stresses are plotted in color, high stress in red, low stress in blue. Nodal displacement can also be enlarged and displayed graphically.



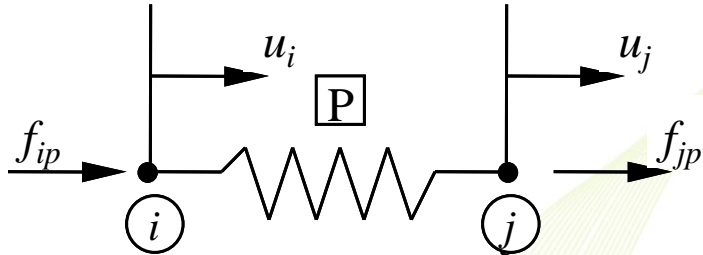
# Building an FEA model -- One-Dimensional Spring



- ✓ Define coordinates of the 3 nodes.
- ✓ Choose the proper type of element considering required nodal degree of freedom from the [element library](#). → There is only 1 DOF in each node in the spring elements used in the model.
- ✓ Assign material properties to the elements. → Spring constant  $k_1, k_2$ .
- ✓ Define the connectivity of the elements. → Geometry of the FEA model is defined by the coordinates of the nodes and connectivity of the elements.
- ✓ Define loading and boundary conditions. → Horizontal force is applied on node 3, and node 1 is constrained.



# A Typical Spring Element



$$\begin{bmatrix} k_p & -k_p \\ -k_p & k_p \end{bmatrix} \begin{Bmatrix} u_i \\ u_j \end{Bmatrix} = \begin{Bmatrix} f_{ip} \\ f_{jp} \end{Bmatrix} \quad (5)$$

$$f_{ip} = k_p (u_i - u_j) \quad (1)$$

$$f_{jp} = k_p (u_j - u_i) \quad (2)$$

$$f_{ip} = k_p u_i - k_p u_j \quad (3)$$

$$f_{jp} = -k_p u_i + k_p u_j \quad (4)$$

$$[\mathbf{k}]\{\mathbf{d}\} = \{\mathbf{f}\} \quad (6)$$

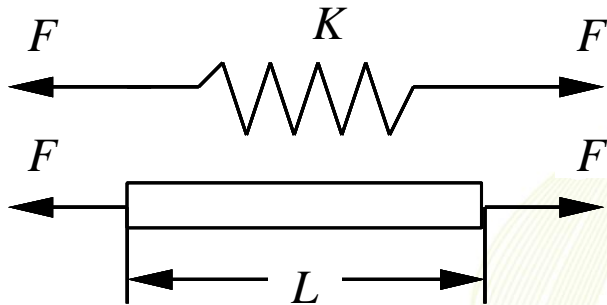
Element  
stiffness matrix

Displacement  
vector

Force vector



# A Truss Element

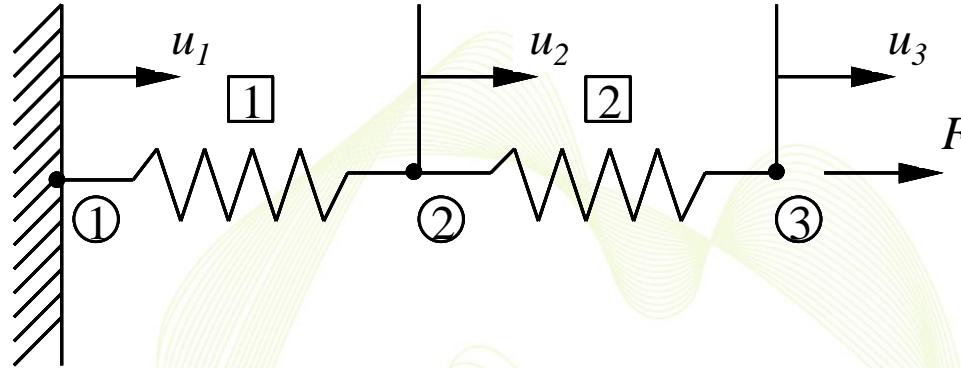


$$[\mathbf{k}] = \begin{bmatrix} \frac{EA}{L} & -\frac{EA}{L} \\ \frac{EA}{L} & -\frac{EA}{L} \end{bmatrix} \quad (11)$$

- ✓ The coordinate of the node (length  $L$ ), element constant (cross sectional area  $A$ ), material property (Young's modulus  $E$ ) are all reflected in the element stiffness matrix.



# Convert the FEA Model into System Equations (I)



✓ For element 1 and 2, Equation (5) becomes

$$\begin{bmatrix} k_1 & -k_1 \\ -k_1 & k_1 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \begin{Bmatrix} f_{11} \\ f_{21} \end{Bmatrix} \quad (7)$$

$$\begin{bmatrix} k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} f_{22} \\ f_{32} \end{Bmatrix} \quad (8)$$





# Convert the FEA Model into System Equations (II)

$$\begin{array}{l} \text{Node 1: } \sum \text{ forces} = 0 \Rightarrow f_{11} = F_1 \\ \text{Node 2: } \sum \text{ forces} = 0 \Rightarrow f_{21} + f_{22} = F_2 \\ \text{Node 3: } \sum \text{ forces} = 0 \Rightarrow f_{32} = F_3 \end{array} \quad \Rightarrow \quad \begin{array}{l} k_1 u_1 - k_1 u_2 = F_1 \\ -k_1 u_1 + k_1 u_2 + k_2 u_2 - k_2 u_3 = F_2 \\ -k_2 u_2 + k_2 u_3 = F_3 \end{array}$$

$$\begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix} \quad (9)$$

$$[\mathbf{K}]\{\mathbf{D}\} = \{\mathbf{F}\} \quad (10)$$

**K**: system stiffness matrix or global stiffness matrix



# Constructing the System Stiffness Matrix

Element 1

$$\begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1+k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix} \quad \begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1+k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} 0 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ 0 \\ F \end{Bmatrix} \quad (12)$$

Element 2

- ✓ In FEA software, components in the element stiffness matrices are “assembled” into proper positions in the system stiffness matrix.
- ✓ The connectivity of the elements are reflected in the system stiffness matrix.
- ✓ The boundary conditions are reflected in the displacement vector and the loading conditions are reflected in the force vector.



# Solving the System Equations

$$\begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} 0 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ 0 \\ F \end{Bmatrix} \quad (12)$$

$$\begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ F \end{Bmatrix} \quad (13)$$

- ✓ In Equation (12), there are 3 linear equations and 3 unknowns. The degree of freedom of node 1 is constrained, therefore Equation (12) can be reduced to Equation (13).
- ✓ Equation (13) can be solved by **Gaussian elimination**, use **forward elimination** to convert the stiffness matrix into a **triangular matrix**, then use **backward substitution** to solve for the unknowns.
- ✓ Solving the system equations is the most computationally intensive process in finite element analysis.
- ✓ **The size of the global stiffness matrix directly relates to the total number of degree of freedom in the finite element model.**
- ✓ For a finite element model with large number of degree of freedom, the cost of computation and storage are high.



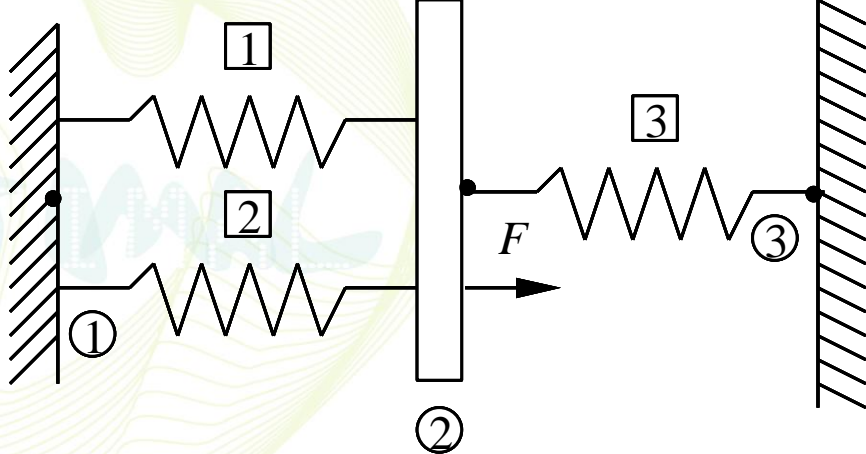
# Example 1. Assemble and Solve the System Stiffness Matrix(I)

✓ Spring constants of the three springs are  $k_1=100\text{N/m}$ ,  $k_2=200\text{N/m}$  and  $k_3=300\text{N/m}$ , respectively. The force applied is 150N. Find the displacement at node 2.

⇒ Construct the element stiffness matrix:

$$[\mathbf{k}] = \begin{bmatrix} k & -k \\ -k & k \end{bmatrix}$$

⇒ Assemble the system stiffness matrix



element1、element2

$$[\mathbf{K}] = \begin{bmatrix} k_1 + k_2 & -k_1 - k_2 & 0 \\ -k_1 - k_2 & k_1 + k_2 + k_3 & -k_3 \\ 0 & -k_3 & k_3 \end{bmatrix}$$

element3



## Example 1. Assemble and Solve the System Stiffness Matrix(II)

$$[\mathbf{K}]\{\mathbf{D}\} = \{\mathbf{F}\}$$

$$\begin{bmatrix} k_1 + k_2 & -k_1 - k_2 & 0 \\ -k_1 - k_2 & k_1 + k_2 + k_3 & -k_3 \\ 0 & -k_3 & k_3 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix} \quad (14)$$

$$\begin{bmatrix} 300 & -300 & 0 \\ -300 & 600 & -300 \\ 0 & -300 & 300 \end{bmatrix} \begin{Bmatrix} 0 \\ u_2 \\ 0 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ 150 \\ F_3 \end{Bmatrix}$$

$$600u_2 = 150 \Rightarrow u_2 = \frac{1}{4} \text{ (m)} \quad F_1 = F_3 = 75\text{N}$$



## Example 2. Calculating Stresses in a Truss Structure (I)

✓  $A_1=A_2=10\text{cm}^2$ ,  $A_3=5\text{cm}^2$ ,  $A_4=4\text{cm}^2$ ,  $F_2=500\text{N}$ ,  $F_3=300\text{N}$ . Find the force and stress in each truss.

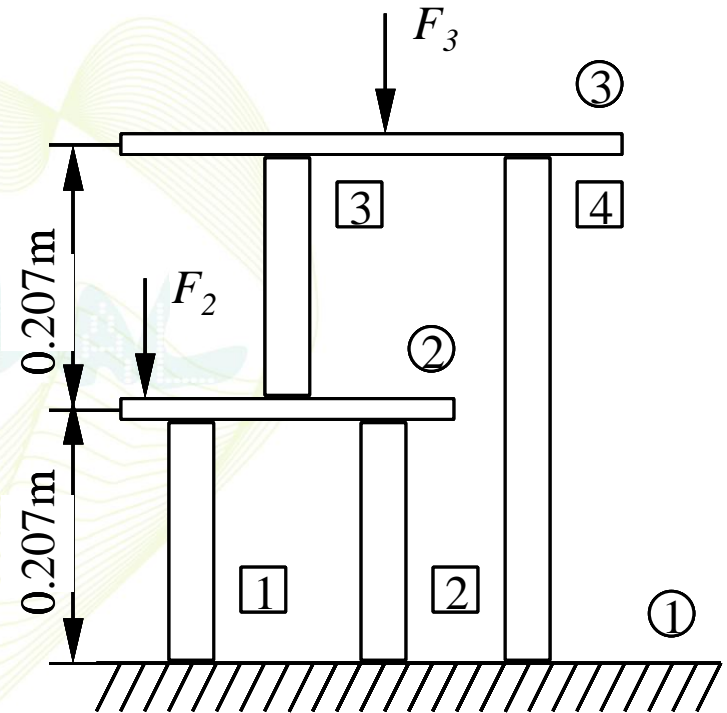
⇒ The element stiffness matrix:

$$[\mathbf{k}] = \begin{bmatrix} \frac{EA}{L} & -\frac{EA}{L} \\ \frac{L}{EA} & \frac{L}{EA} \end{bmatrix}$$

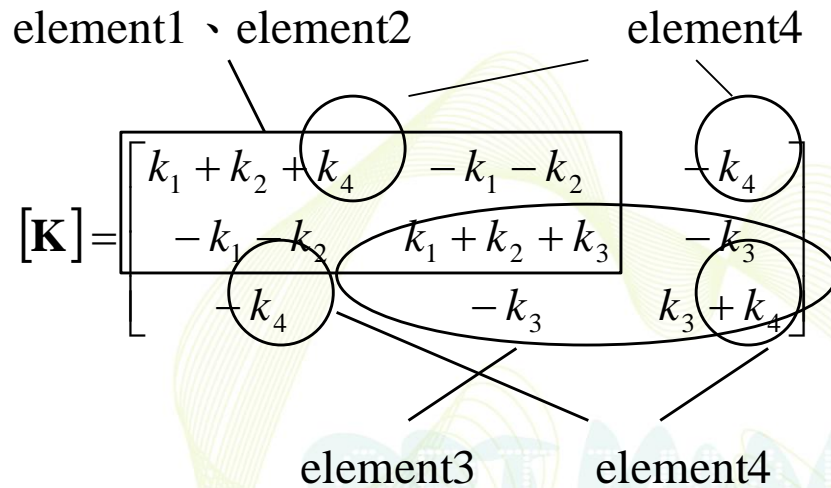
$$[\mathbf{k}_1] = [\mathbf{k}_2] = \begin{bmatrix} 1000 & -1000 \\ -1000 & 1000 \end{bmatrix} \times 10^6 \text{ N/m}$$

$$[\mathbf{k}_3] = \begin{bmatrix} 500 & -500 \\ -500 & 500 \end{bmatrix} \times 10^6 \text{ N/m}$$

$$[\mathbf{k}_4] = \begin{bmatrix} 200 & -200 \\ -200 & 200 \end{bmatrix} \times 10^6 \text{ N/m}$$



## Example 2. Calculating Stresses in a Truss Structure (II)



$$10^6 \times \begin{bmatrix} 2200 & -2000 & -200 \\ -2000 & 2500 & -500 \\ -200 & -500 & 700 \end{bmatrix} \begin{Bmatrix} 0 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ 500 \\ 300 \end{Bmatrix} \quad (15)$$

$$u_2 = \frac{1}{3} \times 10^{-6} (\text{m}) \quad u_3 = \frac{2}{3} \times 10^{-6} (\text{m}) \quad F_1 = -800 (\text{N})$$



## Example 2. Calculating Stresses in a Truss Structure (III)

✓ Strain of truss 1: 
$$\varepsilon_1 = \frac{(u_2 - u_1)}{L_1} = 1.61 \times 10^{-6}$$

✓ Stress of truss 1: 
$$\sigma_1 = E\varepsilon_1 = 0.333(\text{MPa})$$

✓ Forces: 
$$f_1 = A_1 \times \sigma_1 = 333(\text{N})$$

✓ Stress and forces produced in truss 3 and truss 4 are: 
$$\sigma_3 = 0.333(\text{MPa}) \quad f_3 = 166(\text{N})$$

$$\sigma_4 = 0.333(\text{MPa}) \quad f_4 = 133(\text{N})$$





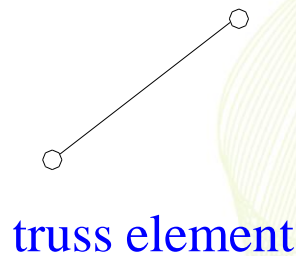
# FEA Applications

- ✓ Messages in Solving Stage of FEA Software
  - Reading inputs
  - Computing element stiffness matrix
  - Assemble system stiffness matrix
  - Begin Gaussian elimination
  - Calculating displacements
  - Calculating stresses
- ✓ Analysis results can be displayed graphically in post-processing stage.
- ✓ Primary considerations when using FEA:
  - Do we really need to do a finite element analysis? Or a fundamental analysis is enough?
  - How much details do we need? Do we need a precise, quantitative analysis, or a conceptual, qualitative analysis?
  - How do we choose proper element types? How do we properly simulate the loading and boundary conditions?

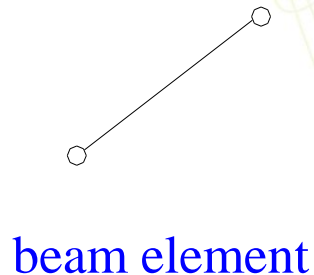


# Common Element Types (I)

- ✓ The most important consideration when choosing element types is the degree of freedom of each node in that element.



- 2 nodes
- DOF of UX and UY
- Only axial load

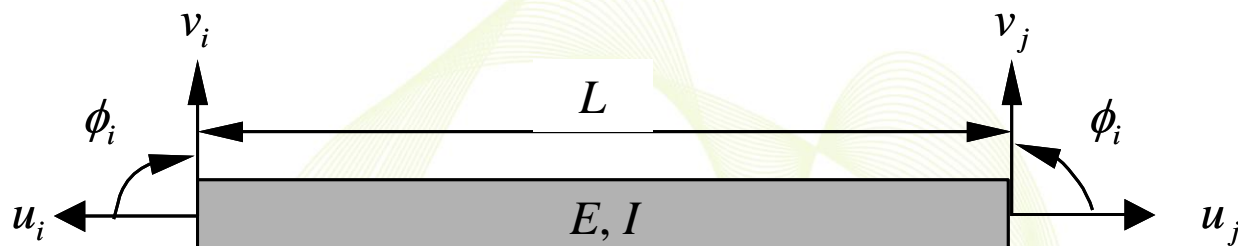


- 2 nodes
- DOF of UX and UY
- DOF of ROTZ
- Can sustain bending



# Beam Element

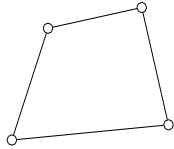
- ✓ In a 2-D beam element, there are three degree of freedom at each node. Therefore a 2-D beam element has a  $6 \times 6$  element stiffness matrix.



$$\mathbf{k} = \begin{bmatrix} \frac{AE}{L} & 0 & 0 & -\frac{AE}{L} & 0 & 0 \\ 0 & \frac{12EI}{L^3} & \frac{6EI}{L^2} & 0 & -\frac{12EI}{L^3} & \frac{6EI}{L^2} \\ 0 & \frac{6EI}{L^2} & \frac{4EI}{L} & 0 & -\frac{6EI}{L^2} & \frac{2EI}{L} \\ -\frac{AE}{L} & 0 & 0 & \frac{AE}{L} & 0 & 0 \\ 0 & -\frac{12EI}{L^3} & -\frac{6EI}{L^2} & 0 & \frac{12EI}{L^3} & -\frac{6EI}{L^2} \\ 0 & \frac{6EI}{L^2} & \frac{2EI}{L} & 0 & -\frac{6EI}{L^2} & \frac{4EI}{L} \end{bmatrix}$$

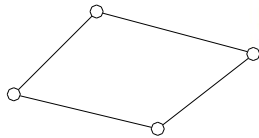


## Common Element Types (II)



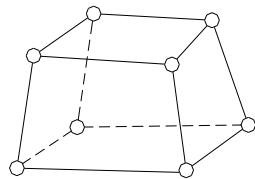
plane-stress element

- 4 nodes
- DOF of UX and UY
- For plane-stress and plane strain analysis



shell solid

- 4 nodes
- DOF of UX, UY and UZ
- DOF of ROTX, ROTY and ROTZ
- For shell and plate analysis

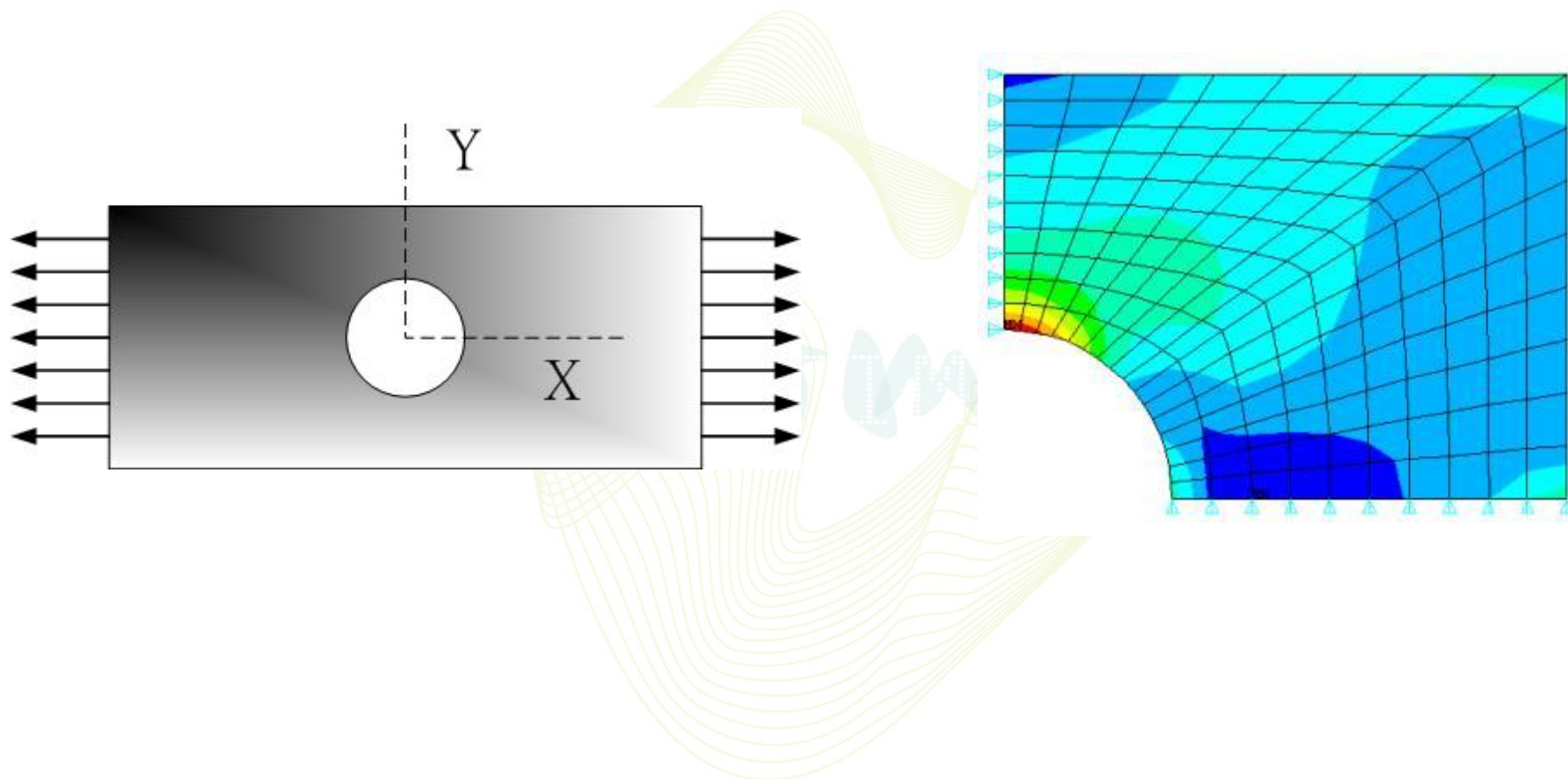


3D solid

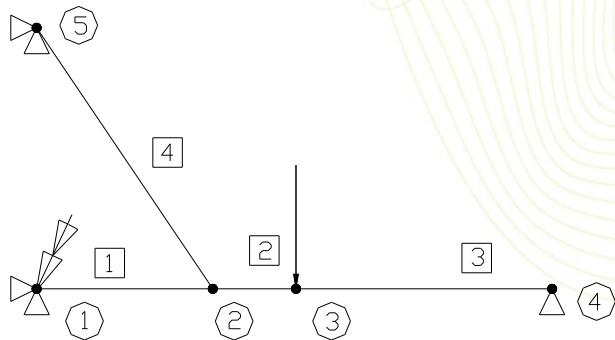
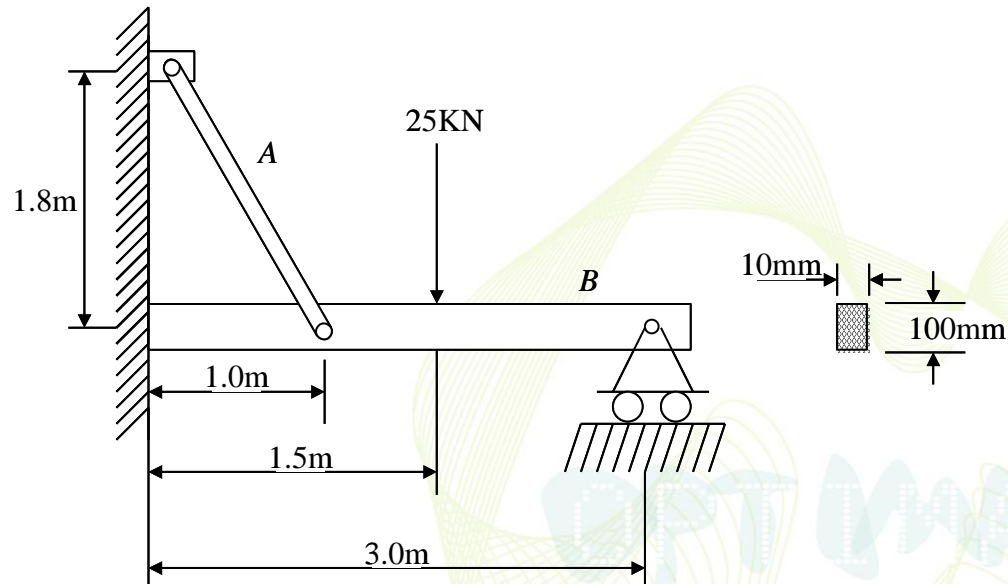
- 8 nodes
- DOF of UX, UY and UZ
- DOF of ROTX, ROTY and ROTZ
- For general analysis



## Example 3. Symmetric Boundary Conditions



# Example 4. Planning an FEA Model

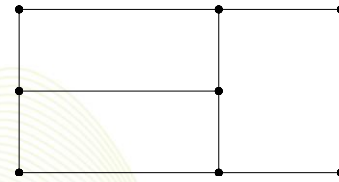
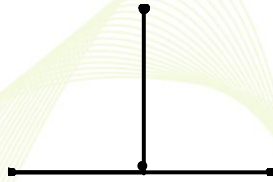
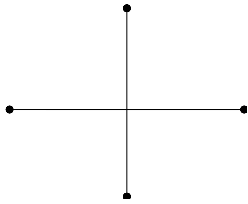


✓ Elements 1, 2, 3 are beam elements, element 4 is truss element.



# Common Mistakes When Using FEA Software (I)

## ✓ Incorrect Connectivity



## ✓ Element distortion



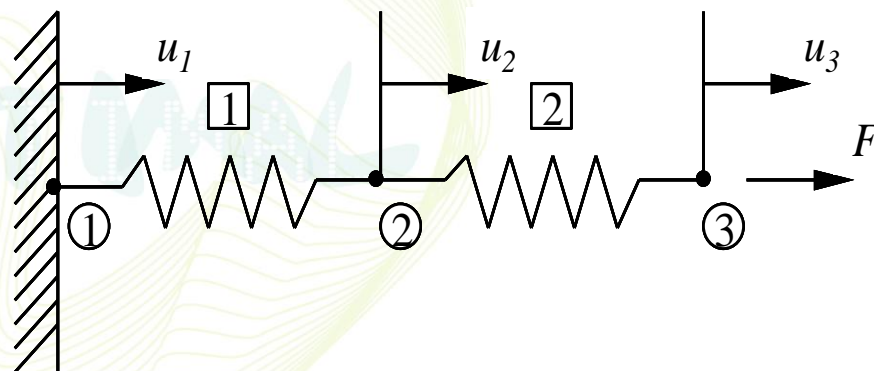
# Common Mistakes When Using FEA Software (II)

- ✓ Insufficient boundary conditions

$$\begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ F \end{Bmatrix} \quad (16)$$

$$\begin{bmatrix} 1 & 0 & -1 \\ 0 & 1 & -1 \\ 0 & 0 & 0 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ F \end{Bmatrix} \quad (17)$$

$$0 \cdot u_3 = F \quad ??$$



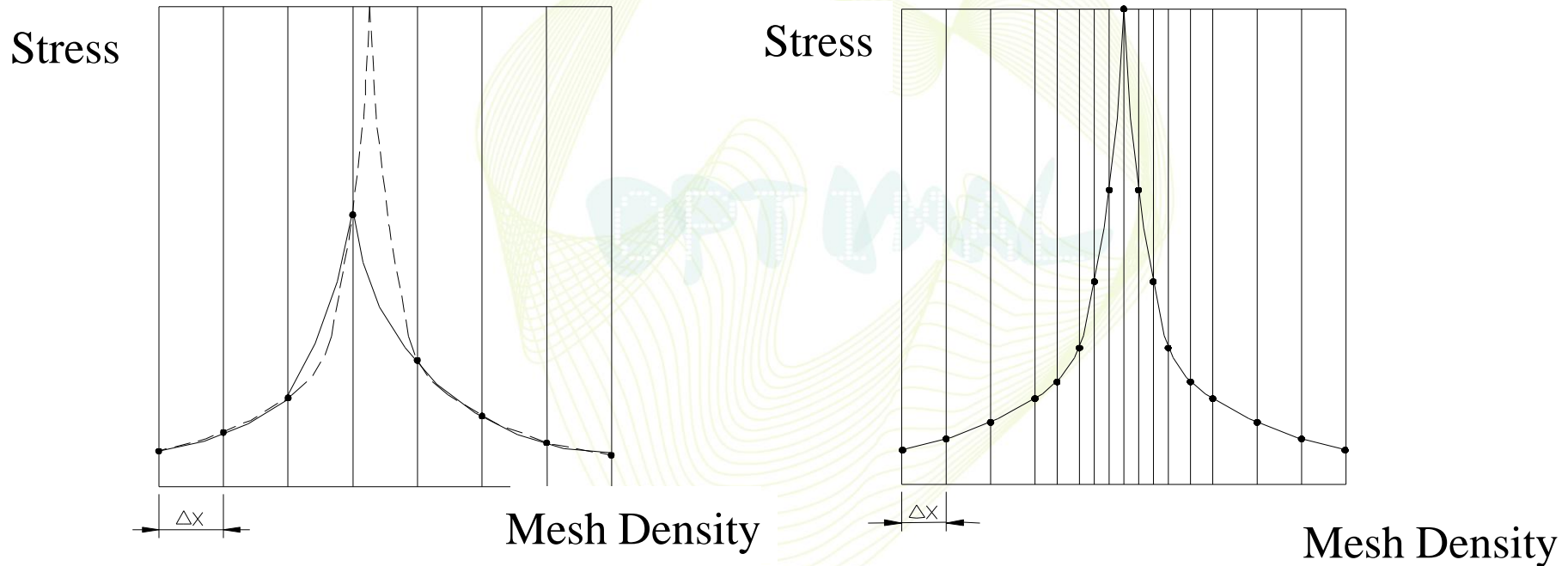
Rigid-body motion would occur if DOF UX at node 1 exists.





# Common Mistakes When Using FEA Software (III)

- ✓ Units have to be consistent.
- ✓ Use higher mesh density in the regions that have higher stress gradient.



## Example 5. Convergence Test to Decide Mesh Density

